

(Derivation:

so tha

A3 (0.4 pt) Find an expression for the distance d as a function of b and the quantities (1). You may also include r_2 and h_2 as variables in your expression, as they will be calculated in subtask A.5.

 $\frac{bMg}{I_S}\sin\varphi \simeq$

 $\frac{bgM}{I_S}$

 $\frac{bgM}{I_S}\varphi$

5.108

[0.4]

0.2

[0.7]

0.2

-0.1

-0.1

 $\mathbf{0.2}$

0.1

0.2

Solution A3:

Some version of the center of mass equation, e.g.

 $b = \frac{dM_2}{M_1 + M_2}$

 $\frac{bM}{\pi h_2 r_2^2 (\rho_2 - \rho_1)}$

correct solution:

A4 (0.7 pt) Find an expression for the moment of inertia I_S in terms of b and the known quantities (1). You may also include r_2 and h_2 as variables in your expression, as they will be calculated in subtask A.5.

Solution A4:

correct answer for moment of inertia of homogeneous disk

$$I_1 = \frac{1}{2}\pi h_1 \rho_1 r_1^4$$

Mass wrong Factor 1/2 wrong in formula for moment of inertia of a disk Correct answer for moment of inertia of 'excess' disk:

$$I_2 = \frac{1}{2}\pi h_2(\rho_2 - \rho_1)r_2^4$$

Using Steiner's theorem:

$$s = I_1 + I_2 + d^2 \pi r_2^2 h_2 (\rho_2 - \rho_1)$$

correct solution:

 $I_S = \frac{1}{2}\pi h_1 \rho_1 r_1^4 + \frac{1}{2}\pi h_2 (\rho_2 - \rho_1) r_2^4 + \frac{b}{\pi r_2^2 h_2}$

In terms of d rather than b gives 0.1pts rather than 0.2pts for the final answer:

$$I_{S} = \frac{1}{2}\pi h_{1}\rho_{1}r_{1}^{4} + \frac{1}{2}\pi h_{2}(\rho_{2} - \rho_{1})r_{2}^{4} + d^{2}\pi r_{2}^{2}h_{2}(\rho_{2} - \rho_{1})$$

A5 (1.1 pt) Using all the above results, write down an expression for h_2 and r_2 in terms of b, T and the quantities (1). You may express h_2 as a function of r_2 .

Solution A5:

It is not clear how exactly students will attempt to solve this system of equations. It is likely that they will use the following equation:

$$M = \pi r_1^2 h_1 \rho_1 + \pi r_2^2 h_2 (\rho_2 - \rho_1) .$$

solve I_S for r_2^2 :

$$r_2^2 = \frac{2}{M - \pi r_1^2 h_1 \rho_1} \left(I_S - \frac{1}{2} \pi h_1 \rho_1 r_1^4 - b^2 \frac{M^2}{M - \pi r_1^2 h_1 \rho_1} \right)$$

replace I_S by T:

$$I_S = \frac{MgbT^2}{4\pi^2}$$

solve correctly for r_2 :

$$r_2 = \sqrt{\frac{2}{M - \pi r_1^2 h_1 \rho_1} \left(M \frac{bgT^2}{4\pi^2} - \frac{1}{2} \pi h_1 \rho_1 r_1^4 - b^2 \frac{M^2}{M - \pi r_1^2 h_1 \rho_1} \right)}$$
0.1

write down an equation for h_2 along the lines of $M = \pi r_1^2 \rho_1 h_1 + \pi r_2^2 (\rho_2 - \rho_1) h_2$ and solve it correctly:

$$h_2 = \frac{M - \pi r_1^2 \rho_1 h_1}{\pi r_2^2 (\rho_2 - \rho_1)}$$

Part B. Rotating Space Station (6.5 points)

B1 (0.5 pt) At what angular frequency ω_{ss} does the space station rotate so that the astronauts experience the same gravity g_E as on the Earth's surface?

Solution B1:

An equation for the centrifugal force along the lines of

 $F_{ce} = m\omega^2 r$

0.1

[0.5]

0.1

[1.1]

0.3

0.1

 $\mathbf{0.2}$



Solution B6:

There are several possible solutions. Solution one – Using Coriolis force

• Velocity v_x

Equation for Coriolis force with correct velocity:

 $F_C(t) = 2m\omega_{ss}^2 Rt\omega_{ss} = 2m\omega_{ss}^3 Rt$

Integrate this, or realize that it is like uniform acceleration for the velocity:

 $t = \sqrt{2H/\omega_{ss}^2}$

 $v_x = 2H\omega$

 $v_x(t) = \omega_{ss}^3 R t^2$

plug in correct value for

overall correct result

The displacement d_x : Integrate $v_x(t)$:

 $d_x = \frac{1}{3}R\omega_{ss}^3 t^3$

Instead of integrating, students may simply 'average' by taking $\frac{1}{2}$ of the final velocity. This gives a factor of $\frac{1}{2}$ instead of $\frac{1}{3}$. Deduct a total of 0.1 pts for this.

Plug in value for t

$$d_x = \frac{1}{3} R \omega_{ss}^3 (2H/\omega_{ss}^2 R)^{3/2} = \frac{1}{3} 2^{3/2} H^{3/2} R^{-1/2} = \frac{1}{3} \sqrt{\frac{8H^3}{R}}$$
 0.2

Solution two – Using inertial frame This solution is similar to the way to solve B7, but needs more complicated approximations than Solution one.

• v_x

Here ϕ denotes the angle swept by the mass and α the angle the astronauts (and tower) has rotated when the mass lands on the floor, see

Initially the velocity of the mass in an inertial frame is $v_x = \omega_{ss}(R - H)$.

When the mass lands, the x-direction has been rotated by ϕ so the new horizontal velocity component is then

$\omega_{ss}(R-H)\cos\phi$

(Student may also write $\cos \alpha$ instead of $\cos \phi$, since $d_x \ll H$.)

$$\cos\phi = \frac{R-H}{R} = 1 - \frac{H}{R}$$

Transforming to the rotating reference frame, one needs to subtract $\omega_{ss}R$. Finally in the reference frame of the astronauts

$$v_x = \omega_{ss} R \left(1 - \frac{H}{R} \right)^2 - \omega_{ss} R \approx \omega_{ss} R \left(1 - 2\frac{H}{R} \right) - \omega_{ss} R = -2\omega_{ss} H$$
 0.2

The sign of the velocity depend on the choice of reference direction, so a positive sign is also correct.

0.3

-0.

[1.1]

0.1

0.2

0.2

0.1

0.1

0.1

0.1

0.1

With the notation from the calculation of v_i

 $= (\alpha - \phi)R$

$$\phi = \arccos\left(1 - \frac{H}{R}\right)$$
$$\alpha = \omega_{ss}t$$

where t is the fall time of the mass, which is given by

$$t = \frac{\sqrt{R^2 - (R - H)^2}}{\omega_{ss}(R - H)}$$

(see solution to B7)

• d_x

Writing $\xi \equiv H/R$ this means

 $d_x = \left[\frac{\sqrt{1 - (1 - \xi)^2}}{1 - \xi} - \arccos(1 - \xi)\right] R$

H

which is a valid end answer to the problem. It is possible, but not necessary, to approximate this for small ξ :

$$\arccos(1-\xi) \approx \sqrt{2\xi} \left(1 + \frac{\xi}{12}\right)$$

which after insertion into the equation for d_x and approximation of small ξ yields the same result as in Solution one:

$$d_x = \frac{2}{3}\sqrt{\frac{2H^3}{R}}$$

If this end answer misses the factor 2/3, deduct 0.1 points.

Solution three – Inertial frame with geometry trick This is an alternative solution to obtain d_x

The mass travels the distance l, and during the fall the space station rotates by ϕ , see Figure 2. According to the intersecting chord theorem,

$$l^2 = H(2R - H)$$

The rotated angle is $\phi = \omega_{ss} t$ where

$$t = \frac{l}{R - H}$$

is the fall time. Thus

$$\phi = \frac{\sqrt{H(2R-H)}}{R-H}$$

 $-\arcsin\frac{l}{R} = \frac{\sqrt{H(2R-H)}}{R-H}$ - $\arctan{\sqrt{x(2)}}$ -0.1

0.1

0.1

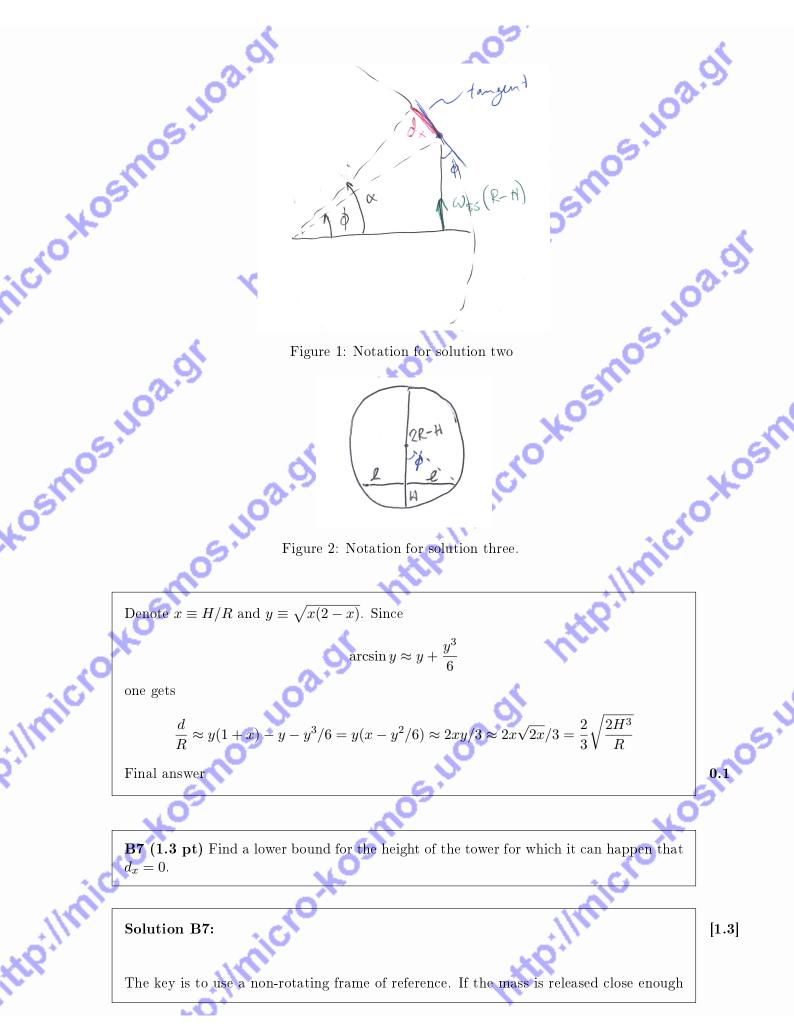
0.1

0.1

0.1

0.1

0.3



to the center, its linear velocity will be small enough for the space station to rotate more than 2π before it hits the ground. The velocity is given by

$$=\omega_{ss}(R-H)$$

distance d that the mass flies before hitting the space station

$$= R^2 - (R - H)^2$$

use non-rotating frame of reference to obtain time t until impact

$$t = d/v = \frac{\sqrt{R^2 - (R-H)^2}}{\omega_{ss}(R-H)}$$

Now there are several possible ways to relate H and the rotated angle ϕ of the space station:

Solution one

$$t = \frac{R\sin\phi}{\omega_{**}R\cos\phi}$$

This time must match $t = \phi/\omega_{ss}$. Obtain the equation

$$\phi = \tan \phi$$

Realizing that there is an infinite number of solutions. This equation has one trivial solution $\phi = 0$, next solution is slightly less than $3\pi/2$ which corresponds to the case H > R (and is thus not correct). The one that gives a lower bound for H is the third solution

$$\phi \approx 5\pi/2$$

The equation $\phi = \tan \phi$ can be solved graphically or numerically to obtain a close value $(\phi = 7.725 \text{ rad})$ which means

$$H/R = (1 - \cos \phi) \approx 0.871$$

Give points if the method is correct, depending on the value of H/R found, according to these intervals:

 $0.85 \le H/R \le 0.88$: 0.4 pts $0.5 \le H/R < 0.85$: 0.3 pts 0 < H/R < 0.5 or H > 0.88: 0.2 pts H = 0 or method is incorrect: 0 pts

Solution two relation between H and rotated angle ϕ

obtain equation of the form

$$\frac{H}{R} = 1 + \cos\left(\frac{\sqrt{1 - (1 - H/R)^2}}{1 - H/R}\right)$$

Figure 3 gives a plot of $f(x) = 1 - \cos\left(\frac{\sqrt{1-(1-x)^2}}{1-x}\right)$. The goal is to find an approximate solution for the second intersection. The first intersection is discarded – it is introduced because of $\cos \phi = \cos(-\phi)$ and corresponds to a situation with H > R. Realizing that there is an infinite number of solutions.

0.2

0.1

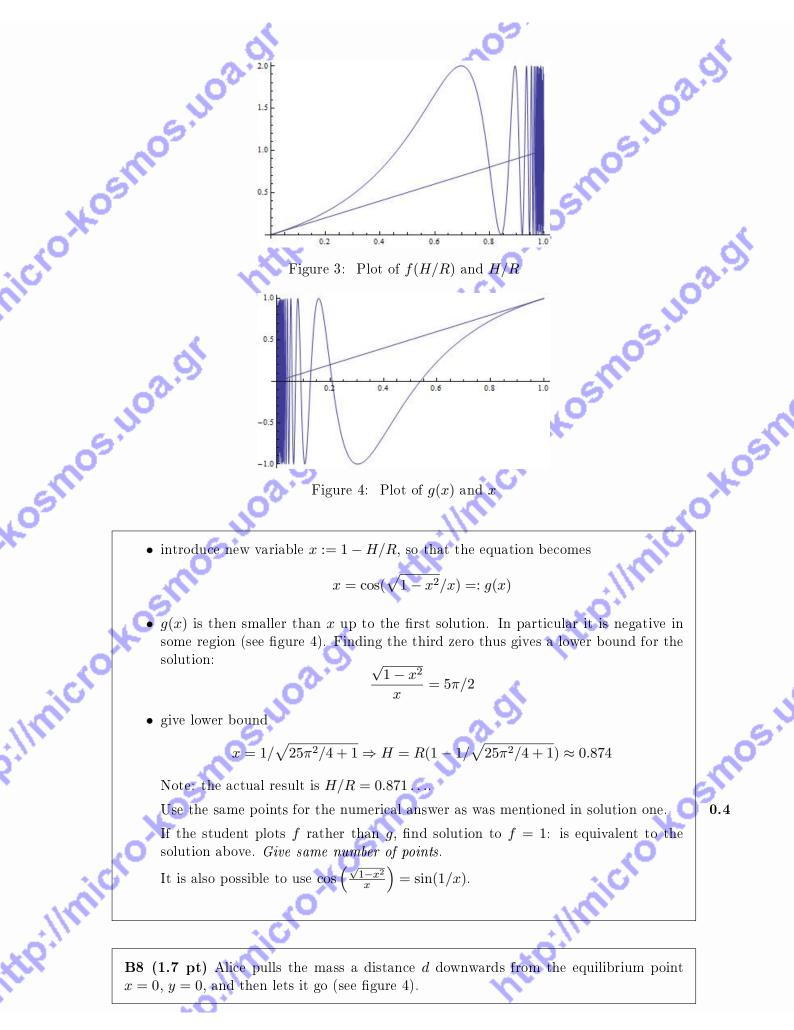
0.1

0.2

0.4

0.2





- Give an algebraic expression of x(t) and y(t). You may assume that $\omega_{ss}d$ is small.
- Sketch the trajectory (x(t), y(t)), marking all important features such as amplitude.

[1.7]

Solution B8:

Note: we did not specify the overall sign of the Coriolis force. Give same amount of points if using opposite convention, but it has to be consistent! Otherwise: subtract 0.1pt for each instance of inconsistency. -0.1Students are allowed to express everything in terms of ω , they don't need to write $\sqrt{k/m - \omega_{ss}^2}$ explicitly. Deduct 0.1pt however if they use k/m instead of ω ... -0.1Realize that y(t) is standard harmonic oscillation: $y(t) = A\cos\omega t + B$ 0.1 Give correct constants from initial conditions $y(t) = -d\cos\omega t$ 0.2Correct expression for $v_y(t)$: $v_y(t) = -d\omega \sin \omega t$ 0.1 Coriolis force in x-direction $F_x(t) = 2m\omega_{ss}v_y(t) = -2m\omega_{ss}d\omega\sin\omega t$ 0.2Realize that this implies that x(t) is also a harmonic oscillation... 0.1 \dots but with a constant movement term superimposed: vt0.1getting the correct amplitude: $\checkmark A = \frac{2\omega_{ss}d}{\omega}$ 0.1 Correct answer with correct initial conditions: $x(t) = \frac{2\omega_{ss}d}{\omega}\sin\omega t - 2\omega_{ss}dt$ 102.01 0.2Sketch: sillmicro.ko ∂d Θ -d 4πω_{ss}d B

